## UNCLASSIFIED

434592 AD 434592

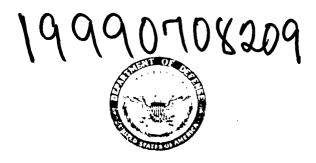
## DEFENSE DOCUMENTATION CENTER

**FOR** 

SCIENTIFIC AND TECHNICAL INFORMATION

CAMERON STATION, ALEXANDRIA, VIRGINIA

Reproduced From



UNCLASSIFIED

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

RPL-TDR-64-38

## A COMPATIBILITY EVALUATION OF PLASTIC MEMBRANE FILTERS WITH LIQUID PROPELLANTS AND SOLVENTS

TECHNICAL DOCUMENTARY REPORT NO. RPL-TDR-64-38

FEBRUARY 1964

AIR FORCE ROCKET PROPULSION LABORATORY
RESEARCH AND TECHNOLOGY DIVISION
AIR FORCE SYSTEMS COMMAND
EDWARDS, CALIFORNIA

CATHLOCKED BY JUC AS HID 1900.



## A COMPATIBILITY EVALUATION OF PLASTIC MEMBRANE FILTERS WITH LIQUID PROPELLANTS AND SOLVENTS

TECHNICAL DOCUMENTARY REPORT NO. RPL-TDR-64-38
February 1964

AIR FORCE ROCKET PROPULSION LABORATORY
RESEARCH AND TECHNOLOGY DIVISION
AIR FORCE SYSTEMS COMMAND
Edwards, California

RPL-TDR-64-38 February 1964

## A COMPATIBILITY EVALUATION OF PLASTIC MEMBRANE FILTERS WITH LIQUID PROPELLANTS AND SOLVENTS

By ROBERT A. BIGGERS

This report has been reviewed and approved.

Colonel, USAF

Director, Air Force Rocket Propulsion Laboratory

### **ABSTRACT**

Data is presented to show the compatibility of polyethylene filter membranes with eleven propellants and eight solvents. More than 300 tests were conducted at ambient temperatures for periods up to 52 hours. The filters tested are satisfactory for use in the gravimetric determination of particulate matter in most liquid propellants.

## TABLE OF CONTENTS

		Page
ABS	STRACT	•
INT	RODUCTION	1
TES	ST PROGRAM	1
	Description of Item Tested	1
	Description of Test Apparatus	1
	Propellants Used	2
	Solvents Used	2
	Number of Tests	3
-	Operational Procedure	3
	Test Results	4
	Effect of Solvents	4
COI	NCLUSIONS	7
вів	LIOGRAPHY	8
	LIST OF TABLES	
I	Compatibility of Membranes with MMH and IRFNA	5
II	Compatibility of Membranes with Solvents	6
API	PENDIX	
	A - Compatibility of Prototype Membranes	A-1
	B - Compatibility of Intermediate Membranes	B-1
	C - Compatibility of Production Membranes	C-1

## A COMPATIBILITY EVALUATION OF PLASTIC MEMBRANE FILTERS WITH LIQUID PROFELLANTS AND SOLVENTS

### INTRODUCTION.

Particle contamination and procedures for measuring particulates have long been serious problems in rocket propulsion systems. To better understand and resolve some of the problems in this area, two principal contracts were awarded: Air Products<sup>(1)</sup> for measuring contaminants in cryogenics, and the Fluor Corporation<sup>(2-5)</sup> to develop methods for sampling and laboratory analysis. One of the procedures recommended by the contractors involved the use of the cellulose membrane filter. However, the cellulose membrane is not compatible with most storable propellants and often distintegrates when used with cryogenics.

A membrane filter was developed by the Millipore Corporation<sup>(6)</sup> to replace the cellulose filter. This new filter is relatively inert to all missile propellants, has good retention efficiency, provides surface filtration, has uniform pore size in the 1-10 micron range, and the weight of an individual membrane is less than 100 milligrams.

The purpose of this report is to present the data obtained in the evaluation of the plastic membrane filter with propellants.

### TEST PROGRAM.

Description of the Item Tested: The membrane filters tested were polyethylene, 47 mm in diameter, approximately 0.3 mm in thickness, and vary in weight 60-100 mg. Two pore sizes were tested: 1.5 and 10 microns.

<u>Description of Test Apparatus</u>: The test apparatus used in the filter evaluation was composed of standard items:

1. All-glass filter apparatus -- 47 mm pyrex filter holder with tapered ground glass joint between holder base and flask, holding clamp, 300 ml funnel and 1-liter ground joint pyrex flask.

- 2. Two-inch diameter glass petri dishes.
- 3. Stainless steel water aspirator.
- 4. Analytical balance with a minimum sensitivity of 0.05 mg.
- 5. Vacuum oven.

Propellants Used: All propellants used for testing purposes were taken from storage drums or tanks. The propellants conformed to the requirements listed in the following military specifications:

- 1. Propellant, Hydrazine unsDimethylhydrazine (MIL-P-27402).
- 2. Propellant, Monomethylhydrazine (MIL-P-27404).
- 3. Propellant, unsDimethylhydrazine (MIL-P-25604A).
- 4. Propellant, Hydrazine (MIL-P-26536A).
- 5. Propellant, Hydrogen Peroxide (MIL-P-16005D).
- 6. Propellant, Nitrogen Tetroxide (MIL-P-26539).
- 7. Propellant, JP-4/UDMH (MIL-P-26694).
- 8. Rocket Fuel RP-1 (MIL-P-25576B).
- 9. Propellant, Nitric Acid (MIL-P-7254 Type III).
- 10. Propellant, Nitrogen, Pressurizing Agent, Type II (MIL-P-27401B).
- 11. Propellant, Normal Propyl-Nitrate (MIL-N-8722A).

Solvents Used: A limited number of solvents were used to evaluate the membrane filters. The following solvents were selected on the basis of their compatibility with various propellants:

- 1. Water
- 2. Acetone
- 3. Methylene chloride
- 4. Petroleum Ether
- 5. Isopropyl Alcohol
- 6. Et. vl Alcohol
- 7. Carbon Tetrachloride
- 8. Chloroform

the polyethylene membrane filters. The membranes are categorized as follows: the prototype, the intermediate, and the production item. Over 130 tests were conducted on the prototype membranes. 110 tests on the intermediate, and 90 tests on the production item. The 1.5 micron and 10 micron membranes are grouped together, although they were tested separately. The manufacturing process differed slightly for filters of the two pore sizes, but no distinct difference was noted regarding each membrane's compatibility with the propellants.

Operational Procedure: Prior to each sample test, all items of the filtration apparatus and sample bottle were cleaned with detergent, rinsed twice with distilled water, rinsed twice with isopropyl alcohol, and allowed to dry. Finally, the apparatus was rinsed twice with filtered petroleum ether and allowed to dry until the petroleum ether vapors completely disappeared.

One 10-micron (or 1.5 micron) solvent resistant filter disc was weighed to the nearest 0.1 mg. Forceps were used to place the filter disc in a covered petri dish. An additional filter was weighed and set aside as the control filter.

The control filter disc was removed from the petri dish with forceps and the filter placed on the filter holder base. The tared filter disc was placed directly on top of the control filter. The ...lter holder funnel was clamped to the base, the vacuum system turned on, and the sample filtered in 200-ml portions. Next, the sample bottle was rinsed with filtered solvent (compatible with the propellant). The sides of the funnel and filter disc were rinsed by pouring the sample bottle rinsings into the funnel. The funnel was disconnected from the base and the filter disc edges rinsed carefully with a jet of filtered solvent to accumulate the particlate residue toward the center of the disc while evacuation continued. The vacuum was released. Forceps were used to place the filter disc in the covered petri dish. The control filter disc was placed in an additional covered petri dish. The dishes, covers ajar, were placed in a vacuum oven at 70°C for 30 minutes. The filter discs were then reweighed at ambient temperature to the nearest 0.1mg.

An additional procedure was used to evaluate the membranes. The tared test membrane and 100 ml of propellant were placed in a closed bottle, mechanically shaken for periods of 15 minutes to 52 hours, and reweighed.

Test Results: The prototype membrane filters were not compatible with most of the propellants. Also, they were unsatisfactory for gravimetric use, i.e., a 100-mg. filter disc lost 10% of its total weight when subjected to red fuming nitric acid (RFNA) for 50 hours in a shaker. These same filters lost 4% of their total weight on direct filtration of 200 ml of RFNA. As the formulation of polyethylene improved, the chemical attack on the membranes by the propellants was less severe.

During the testing of the prototype filters, RFNA and monomethyl-hydrazine (MMH) were found to be the most incompatible of the propellants used. MMH and RFNA were used almost exclusively to evaluate the intermediate and production item membranes. Typical test results with these two propellants are shown in Table I.

The intermediate filters lost approximately two mg (100 mg disc) with the long-duration RFNA tests, and the production item lost one mg. when subjected to the same test. (A complete list of data is included in the Appendix.)

No membrane completely failed. Swelling occurred with the hydrocarbon fuels, but the membranes showed no loss of elasticity and no distortion with any of the other propellants.

Effect of Solvents: The solvents were more compatible with the production membranes than were the propellants. The results of a 16-hour shaker test are reported in Table II.

The solvent extraction is of little importance, since the initial weight loss is accomplished by the propellant filtration. The tests indicate that the maximum weight loss of membranes occurs in the initial contact with the propellant.

COMPATIBILITY OF MEMBRANES WITH MMH AND RFNA

Control		Run #1	Weight	Run #2	Weight	Run #3	Weight
Membrane	ne Initial	Prop.	Loss	Prop.	Loss	Prop.	Loss
No.	-1	Filtered	mg.	Filtered	mg.	Filtered	mg.
Proto #2	106.7	200 ml. MMH	9.1	200 ml. MMH	1.5	200 ml. MMH	0.3
Proto #10	.0 98.2	200 ml. RFNA	10.4	200 ml. RFNA	8.0	200 ml. RFNA	0.0
Inter. #1	1 76.5	200 ml. MMH	₩.0	200 ml. MMH	₩.	200 ml. MMII	0.1
Inter. #2	2 77.6	200 ml. RFNA	0.5	200 ml. RFNA	0.1	200 ml. RFNA	0.0
Prod. #1	1 95.3	500 ml. MMH	0.1	500 ml. MMH	0.0	500 ml. MMH	0.0
Prod. #7	7 99.3	.500 ml. RFNA	0.7	500 ml. RFNA	9.4	500 ml. RFNA	0.0

TABLE II
COMPATIBILITY OF MEMBRANES WITH SOLVENTS

Control Membrane No.	Initial Weight mg	Solvent in Shaker	Weight Loss mg.
Sol. 1	102.1	Methylene Chloride	0.5
Sol. 11	98.0	Petroleum Ether	0.5
Sol. 13	99.7	Acetone	0.3
Sol. 15	101.4	Chloroform	0.9
Sol. 17	100.1	Carbontetrachloride	0.9
Sol. 19	99.1	Isopropyl Alcohol	0.1
Sol. 21	97.5	Ethyl Alcohol	0.1

## CONCLUSIONS.

The production membranes, now marketed as Type OS and OH filters, are resistant to most propellants. The membranes are compatible with all of the solvents tested. The membranes may not be used for gravimetric analysis with hydrocarbon-type fuels.

The membrane filters are inexpensive, disposable, easy to use, and have excellent chemical resistance.

## **BIBLIOGRAPHY**

- 1. Foster, R.H., "Study of Liquid Contamination," Air Products and Chemicals, Inc., SSD-TDR-62-8, May 1961.
- 2-5. Dutch, Paul H., MacInnes, W.F., "Development of Sampling Techniques and Particulate Matter Analysis," the Fluor Corporation, Ltd., AFFTC-TR 60-44, 60-45, 60-46, 60-47, November 1959.
- 6. "Detection and Analysis of Contamination," Millipore Filter Corporation ADM-30, November 1962.

## APPENDIX A

COMPATIBILITY OF PROTOTYPE MEMBRANES

## COMPATIBILITY OF PROTOTYPE MEMBRANES

		-							
Percent Total Loss	10.8	10.3	12.0	7.6	12.0	9.1	10.3	9.2	10.8
Wt. Loss	0.0	0.0	!	i	0.0	0.1	0.0	0.1	l
Run #5 Prop. Test	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml.	I	l	200 ml. Mai	200 ml.	200 ml. MM	200 ml. MMH	I
Wt. Loss	0.5	0.1	0.0	••	4.0	0.3	0.1	0.2	0.0
Run #4 Prop. Test	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml.	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml. 0.4 Mil	200 ml.	200 ml.	200 ml. MM	200 ml. MMH
Wt. Loss	0.3	0.3	0.1	0.0	0.0	0.0	0.3	0.0	0.1
Rum #3 Prop. Test	200 ml. RFNA	200 ml. MMH	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml JP-X	200 ml. JP-X	200 ml. MM	200 ml. N-Propyl-	200 ml. Mai
Wt. Loss	1.7	1.5	4.0	0.5	0.0	0.0	0.0	0.0	0.2
Run #2 Prop. Test	200 ml N <sub>2</sub> H <sub>4</sub>	200 ml.	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml. N <sub>2</sub> H4	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml. N <sub>2</sub> H <sub>4</sub>	200 ml. H.D. (90Z)	200 ml RFNA	200 ml. N <sub>2</sub> H <sub>4</sub>
Wt. Loss	9.1	9.1	4.6	æ .3	11.5	7.5	9.0	7.7	10.2
Run #1 Prop. Test	200 ml. UDMH/ N <sub>2</sub> H <sub>4</sub>	200 ml. MMH	24-hr Shake N <sub>2</sub> H <sub>4</sub>	24-hr. Shake UDMH/ N <sub>2</sub> H <sub>4</sub>	52-hr Shake N <sub>2</sub> H <sub>4</sub>	52-hr. Shake N <sub>2</sub> H 4	52-hr. Shake N <sub>2</sub> H <sub>4</sub>	52-hr. Shake N <sub>2</sub> H <sub>4</sub>	Shake RFNA
Initial Wt. in mgs.	107.0	106.7	82.6	8*06	99.2	86.1	91.0	85.9	(52-hr 97.4
Control Membrane No.	-	7	m	4	'n	<b>v</b> 9	_	<b>&amp;</b>	6

1		ı									
	Percent Total	Loss	11.4	11.4	<b>&amp;</b>	6°8	o <b>.</b> 6	<b>7.</b> 0	<b>5.</b> 8	6.2	6.3
	¥t.	18.e	1.	1	1	i	1	0.0	0.0	0.0	1
	•	Test	i	1	1	i I	1	200 ml. 0.0 RFNA	100 ml. 0.0	200 ml.	1
	¥t.		0.0	0.0	•	0.0	0.0	0.0	0.0	2.0	0.0
		Test	200 ml. MMH	200 ml.	1	200 ml. MM	200 ml. Mrti	200 ml. Mydd	200 ml.	200 ml.	200 ml.
	WE	Loss ⊞	0.0	0.0	0.0	0.2	0.2	0.0	0.2	<b>9.</b> 0	0.1
	Run #3	Prop. Test	200 ml. RFNA	4-hr. RFNA	200 ml	4-hr. Shake N <sub>2</sub> H <sub>4</sub>	4-hr. Shake N,H	18-hr Shake N,H <sub>k</sub>	18-hr Shake N <sub>2</sub> H <sub>4</sub>	200 ml. Conc. HF	1-hr. Shake N <sub>2</sub> H <sub>4</sub>
	Wt	Loss mg.	0.8	9.0	0.5	0.1	0.5	0.3	0.1	0.2	0.5
	Run #2	· Prop.	200 ml. RFNA	18-hr RFNA	200 ml. N <sub>2</sub> H <sub>4</sub>	18-hr Shake N <sub>2</sub> H <sub>4</sub>	18-hr. Shake N,H	6-hr. Shake N, H,	6-hr Shake UDMH/ N <sub>2</sub> H <sub>4</sub>	100 ml. H <sub>2</sub> O/ Acetone	0.5 hr. Shake N <sub>2</sub> H <sub>4</sub>
	W.	Loss	10.4	10.8	7.0	7.6	7.0	5.3	4.3	0.2	4.3
	Run #1		200 ml.	Li al	42-hr. Shake N <sub>2</sub> H <sub>4</sub>	50-hr. Shake N <sub>2</sub> H <sub>4</sub>	50-hr. Shake N <sub>3</sub> H,	2 4 18-hr. Shake N <sub>2</sub> H <sub>2</sub>	.2 18-hr. Shake UDKHMN <sub>2</sub> H	Heat 170°F	1.5 hr. Shake N <sub>2</sub> H <sub>4</sub>
	Intetal	Wt. in	98.2	9*66	85.6	88.4	85.1	80.2	79.2 UDPA	86.7	78.0
	1000	Controt Membrane	No.	11	12	13	14	15	16	11	18

COMPATIBILITY OF PROTOTYPE MEMBRANES (Cont'd)

_															!
Darcont	Total	Loss	4.7	4.6	6.0			· ·	• •	· ·	0.7	6.4	4.00	1.1	10.6
4	Loss	mR.	1	•	i	}			:	!	1		1	1	1
20.00	Prop.	Test	1	i		}	<b>!</b> .	1	i	1	1	ı	ı	i	1
	Wt. Losa	mg.	0.0	1	1	1	1	1	1	1	1	1	1	1	0.0
	Run #4 Prop. 1	_	200 ml. Per			1	1	I	1		1		ı	1	200 ml.
	Wt.	mg.		ı		i	1	1	1	l	0.1	0.0	0.0	0.0	0.6
	<b>~</b>	Test.	1-hr. Shake RFNA	1		1	1	1	1	1	200 ml. MAGH	200 ml.	200 ml.	200 ml.	200 ml.
	¥t.	ng.	0.1	1		1	1	1	9.0	1	0.7	9.0	0.5	0.5	0.9
	7	Prop. Test	0.5-hr. Shake	Kray		1	1	1	200 ml. RFNA	1	200 ml. MMH	200 ml.	200 ml.	200 ml.	0.5 hr. Shake RFNA
	¥.	Loss	5.9	3,9		0.1	6.9	3.5	4.5	0.1	5.0	3.0	8.5	5.9 If	9.0
	Run f.		ı i.	RFNA	90Z H <sub>2</sub> 0 <sub>2</sub>	200 ml. JP-X	200 ml.	200 ml.	200 ml.	2000 ml. Liq. N <sub>2</sub>	0.1-hr. RFNA	0.2-hr. RFNA	0.5-hr. Shake N <sub>2</sub> H <sub>4</sub>	0.5-hr. Shake UDMH/N <sub>2</sub> H <sub>4</sub>	0.5-hr. Shake RFNA
	Initial	Wt. fa	82.7	6	1	78.4	98.3	83.5	78.4	9.68	82.8	73.5	92.5	83.6	98.6
	Control	Membrane Wt. in	No.	9	<b>3</b> .	21	22	23	24	25	26	27	28	29	30

**A**-3

COMPATIBILITY OF PROTOTYPE MEMBRANES (Cont'd)

					-					•		
		1	15.	Run #2	Wt.	Rum #3	Wt.	Rum #4	Wt.	Kun #5	Mr.	Percent
	Initial	Kun #1	1058	Prop.	Loss	Prop.	Loss	Prop.	lo sa	Prop.	Loss	Total
е е	MC. 111		E	Test	ER.	Test	田路	Test	mg.	Test	E E	Loss
i	31 91.5	0.5-hr. Shake	2.0	0.5-hr. Shake	1.0	200 ml. MMH	5.2	200 ml. Mrti	0.0	1	1	0.6
	64.4	24-hr. Shake	2.2	48-hr. Shake	0.1	ŀ	1	l	ı	1	1	3.6
	72.9	E .	8	HMW.	0.1	ì	i	1	ì	i	1	5.3
	82.8	:	6.3	2	0.1	1	1	1		;	1	7.7
	95.1	=	10.0	=	0.3	ı	ı	ł	1	i	i	10.8
	61.9	=	1.9	2	0.2	i	l	l	1	:	:	3.4
	71.0	=	3.3	=	0.1	1	1	ł	1	ı	I	4.8

## APPENDIX B

COMPATIBILITY OF INTERMEDIATE MEMBRANES

## COMPATIBILITY OF INTERMEDIATE MEMBRANES

Percent Total Loss	1.2	0.7	1.0	1.4	6.0	0.8	0.8	1.1	1.2	1.0	1.2	1.1	2.5	2.6	2.5
Loss BR.	1	1	1	1	ŀ	i	1	1	1	1	1	ı	0.3	0.5	0.5
Run #5 Prop.	1	i	I	:	:	1	1	1	1	i	1	1	0.1-hr. Shake	HAH =	=
Vt. Loss	1	1	1	•	1	1	1	1	1	1	0.0	0.0	9.0	0.5	9.0
Run #4 Prop. Test	1	1	ı	i	i ·	1	- 1	1	i	1	200 ml. Mrti	*	=	:	2
Wt. Loss	0.1	0.0	0.0	0.0	ì	1	1	ı	1	ı	0.2	0.2	0.5	0.5	0.4
Run #3 Prop. Test	200 ml.	200 ml. RFNA	200 ml. MM	200 ml. MM	1	1	1	i	1	1	200 ml. MMH	=	=	2	
Wt. Loss mg.	4.0	0.1	0.1	0.1	1	1	1	0.3	0.2	0.1	0.1	0.1	0.1	0.2	0.1
Run #2 Prop. Test	200 ml.	200 ml. RFNA	200 ml.	200 m1. MPGH	1	:	1	200 ml. N <sub>2</sub> H <sub>4</sub>	<b>.</b>	z	300 ml.		200 ml. N <sub>2</sub> H <sub>4</sub>	=	2
Ht. Loss	9.0	0.5	0.7	1.0	0.1	9.0	9.0	9.0	9.0	9.0	9.0	9.0	0.7	0.7	0.7
Run #1 Prop. Test	200 ml. MMH	200 ml. RFNA	3-hr. Shake N <sub>2</sub> H <sub>4</sub>	3-hr. Shake RPNA	200 ml.	ı	=	200 ml. N <sub>2</sub> H <sub>4</sub>	=	=	200 ml. UDMH/N <sub>2</sub> H <sub>4</sub>	=	200 ml. M <sub>2</sub> H <sub>4</sub>	=	=
Initial Wr. in mgs.	76.5	77.6	79.6	78.3	79.8	72.0	76.2	78.7	68.2		75.0	79.7	89.5	91.5	90.9
Control Membrane No.	1	2	en '	4	٧n	9	7	<b>ω</b>	6	10	11	12	13	14	. 15

			İ															
	Percent	Total Loss		7.7	2.3	1.7	2.1	0.7	8.0	1.0	1.0	1.0	0.7	1.0	0.3	0.3	0.3	7.0
	¥t.	Loss mg.		i	1	1	i	;	ı	i	1	i	:	!	ı	1	1	1
	Run #5	Prop. Test		1	1	1	1	!	1	i	ł	:	1	ł	1	ł	1	1
	¥t.	Loss mg.		0.0	0.0	1	1	•	i	1	l	1	ı	0.0	- 1	ı	ŀ	i
	<b>.</b>	Prop.		200 ml. MMH	=	1	1	ı	1	ı	1	1	1	400 ml. M.H	1	ł	ł	1
	Wt.	Loss		0.1	0.1	1	ŀ	ł	1	ł	l	ı	1	0.1	4	I	1	1
	Run #3	Prop.	1634	200 ml. MM	=	1	ł	1	1	1	ı	ı	1	500 ml. UDM/N <sub>2</sub> H <sub>4</sub>		ı	i	1
	¥r.	Loss	ä	1.3	6.0	0.0	0.0	0.0	0.0	0.3	0.3	0.0	0.0	0.4	0.0	0.0	0.0	0.0
	Run #2	Prop.	Test	200 ml. MMH	:	200 ml. Mrti	=	800 ml.	=	500 ml. UDMH/N <sub>2</sub> H4	ı E	400 ml.	500 ml.	500 ml.	18-hr. Shake	=	800 ml.	:
•	Wr.	Loss	Eg.	0.8	1.0	1.3	1.6	0.5	0.5	0.3	0.3	9.6	0.5	0.1	0.3	0.3	0.3	0.4
	Run #1	Prop.	Test	3-hr. Shake RFNA	=	20-hr. Shake	2	800 ml.	=	500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>	E	20-hr. Shake MMH	500 ml.	500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>	2-hr. Shake MM	r	800 ml.	£
	Tairtal	Wt. in	mgse	98.0	87.5	74.5	78.0	68.8	67.8	59.9	58.6	61.9	61.4	62.4	102.8	101.0	97.4	<b>9.</b> 66
	i	Membrane Wt. in	No.	16	17	81	19	20	21	22	23	24	25	26	27	28	29	30

COMPATIBILITY OF INTERMEDIATE MEMBRANES (Cont'd)

Percent	Total	Loss	0.3	0.3	0.7	8.7	· ·			n w	n r	•	æ "	6.0	1.0	8.0	0.9	1.1
Wt.	Loss	a	i	ı	i	ł	ı			! <b>;</b>	1 1	<b>!</b>		ŀ	ŀ	1	ı	1
Run #5	Prop.	Test	1	1	ŀ	}	ı	ı	1	1		. 1		ŀ	1	1	1	1
Wr.	Loss	mg	1	ŀ	1	l	1	!	ł	i	1	ł	1		I	!	l	1
Run #4	Prop.	Test	1	ı	1	1	1	i	ł	!	1	1	i	į	i	1	1	1
Wr.	Loss	ER.	0.1	0.2	1	1	i	1	1	ł	ł	ł	1	1	}	1	1	1
Run #3	Prop.	Test	800 ml.	800 ml. Men	1	1	1	i	ŀ	ŀ	i	1	1	ł	ı	ľ	i	
HE.	Loss	E.S.	0.0	0.0	1	ł	1	1	ł	}	1	1	ŀ	ł	{	}	1	1
Run #2	Prop.	Test	500 ml. UDMH/N <sub>2</sub> H4	0.1 500 ml. UDMH/N <sub>2</sub> H <sub>&amp;</sub>		1	t	1	1	1	1	1	i	ı	ı		ŀ	
Wt.	Loss	EB.	0.2	0.1		1.3	1.2	0.3	0.3	0.5	0.7	0.8	0.9	1.0	<b>«</b>		<b>6.</b> 0	;
Run #1	Prop.	Test	500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>	500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>	24-hr. Shake MMH	=	2	=	=	170-hr.	Shake MMH	=	2	=	=	=	-	
Initial	Wt. in	mgs.	97.0	96.2	68.9	73.3	75.0	98.9	7.96	93.7	8.46	92.6	96.5	97.0	99.8	99.2	2 60	0.7.6
Control	Membrane	No.	31	32	33	34	35	36	37	38	39	07	41	42	43	77	45	

## APPENDIX C

COMPATIBILITY OF PRODUCTION MEMBRANES
(PROPELLANT TEST & SOLVENT TEST)

## COMPATIBILITY OF PRODUCTION MEMBRANES

Percent Total Loss	0.1	0.0	0.0	0.1	0.1	0.1	1.2	1.2	1.3	1.2	0.9	1.0	0.9	1.0	0.7	0.7	
Wt. Loss mg.	1	1	1	;	1	:	1	:	1	1	0.0	0.1	0.1	0.1	1	1	
Run #5 Prop. Test																ł	
Wt. Loss mg.	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.3	0.3	0.3	0.3	ı	1	
Run #4 Prop. Test	500 ml.	:	:	500 ml.	=	:	SOO ml. RFNA			=	100 ml. RFNA	=	=	=	ı	}	
Wt. Loss																ı	
Rum #3 Prop. Test	500 ml.	2	2	500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>		E	SOO ml. RFNA	=	=	2	100 ml.	; , =	=	I	ı	ł	
Wt. Loss mg.	0.0	0.0	0.0	0.0	0.0	0.0	0.2	0.3	0.2	0.2	0.0	0.1	0.1	0.1	0.1	0.1	
Run #2 Prop. Test	500 ml.		*	.1 500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>	: :		SOO ml. RFNA	2	2	2	100 ml. Pet.Ether	E	=	E	4-hr. Shake RFNA	2	;
Wt. Loss	0.1	0.0	0	0	0	0	0	9.0	0.7	9.0	4.0	0.4	0.3	0.3	9.0	9.0	,
Run #1 Prop.	500 ml.	z	=	500 ml. UDMH/N <sub>2</sub> H <sub>4</sub>		=	500 ml. RFNA	E	=	2	100 ml. CH <sub>2</sub> Cl <sub>2</sub>	=	=	=	70-hr. Shake RFNA	=	=
Initial Wt. in	95.3	95.1				95.4		99.3	100.5	101.8	101.5	100.7	99.5	98.8	8.96	97.0	,
Control Membrane	1	2	m	4	'n	9	_	<b>∞</b>	ת	10	=	12	13	14	15	16	17

# COMPATIBILITY OF PRODUCTION MEMBRANES (Cont'd)

Percent Total Loss	0.1	0.2	0.1	0.1	0.1	0.2	0.2	0.2		1.1	1.1	1.2	1.1	1.2
Wt. Loss	1	1	1	I	1	1	i i	1	ני	1	ļ	!	ŀ	1
Run #5 Prop. Test	1	1	l	ŀ		!	I	ł	Run #5 Solvent Test	1	ı	ł	ł	!
Wt. Loss mg.	1	1	1	1	•	i	1	1		1	l	1	1	1
Run #4 Prop. Test	1	1	1	1	1	1	ŀ	ŀ	Rum #4 Solvent Fest	1	1	I	ı	1
Wt. Loss		l	1	1	1	1	1	1		0.0	0.0	0.0	0.0	0.0
Run #3 Prop. Test	1	1	1	1	ł	1	1	ŀ	Run #3 Solvent Test	4-hr. RFNA	<b>E</b>	=	=	=
Wt. Loss	1.	1	1	1	1	ı	1	l		9.0	9.0	0.5	9.0	9.0
Run #2 Prop. Test	1	1	1	1	1	1	!	1	Run #2 Solvent Test	60-hr. RFNA	:	=	=	=
Wt. Loss	0.1	0.2	0.1	0.1	0.1	0.2	0.7	0.2		0.5	0.5	0.7	0.5	9.0
Run #1 Prop. Tef:	28-hr. Shake N <sub>2</sub> 0,08%	: :	28-hr. Shake N <sub>2</sub> 0 <sub>4</sub> 212	r E	28-hr. N <sub>2</sub> 0 <sub>4</sub> 36%	=	130-hr. N <sub>2</sub> 0 <sub>4</sub> 26%	<b>1</b>	Run #1 Solvent Test	24-hr. CH <sub>2</sub> Cl <sub>2</sub>	î Î =	E	=	£
Initial Wt. in	ا ب	103.3	103.7	100.5	104.6	103.5	101.2	103.8		102.1	103.1	101.7	102.1	104.3
Control Membrane	18	19	20	21	22	23	<b>7 7 7 7</b>	25		-	2	m	4	'n

COMPATIBILITY OF PRODUCTION MEMBRANES (Cont'd)

#5 Wt. Percent nt Loss Total mg. Loss	n1. 0.0	0.1 0.6	0.0 0.6	0.0 0.5	0.0 0.6	0.5	9.0	0.3	0.3	6.0	6.0	0.0		1.0				
Wt. Run #5 Loss Solvent mg. Test	• •	0.1 "	0.1 "	0.1 "	0.3 "	1	1	1	!	i	1	1		1				
Run #4 W Solvent Lo	। । तं ुर	:		:	:	:	1	1	1		1	1		ı				
Wt. Loss	0.0	0.0	0.1	0.1	0.1	i		ŀ	l	i	i	1		1	1 1	111	1111	1111
Run #3 Solvent	100 ml. Pet.Ether	I	:	*	=	1	1	1	i	i	1	1		ı	1 1	111	11 11	1111
Wt. Loss	0.0	0.0	0.0	0.0	0.0	ŧ	1	1	1	ı	i	1		1	1 1	111	11 11	11 11
Run #2 Solvent	100 a1.	:	2	z	=	ŧ	1	1	ł	ŧ	1	ł		:	1 1	1 1 1	1 1 1 1	11 11
Wt.	0.3	0.4	0.4	0.3	0.3	0.5	9.0	0.3	0.3	0.9	0.9	0.9		1.0	1.0	1.0	1.0 0.1 0.1	1.0 0.1 0.1
= =	100 ml.	I			2	6-hr. Shaker Pet Ether	2	16-hr.	n = n	16-hr CHCl,	n =	16-hr.	7400	4 :	16-hr.	16-hr.	16-hr. 16-hr. 16-hr.	16-hr. 16-hr. 16-hr. 16-hr.
Initial Wt. in	100.1	101.5	100.7	99.5	98.8	98.0	98.2		100.1	101.4	101.3	100.1		99.8	99.8	99.8	99.8	99.8
Control Membrane	NO.	7	∞	6	10	<b>:</b> .	12	<b>E</b>	14	15	16	11		18	18	18 19 20	18 19 20 21	18 19 20 21

COMPATIBILITY OF PRODUCTION MEMBRANES (Cont'd)

Control Membrane No.	Initial Wt. in Smgs.	Run #1 Solvent Test 24-hr. Shake	Loss 18.	Run #2 Solvent Test 70-hr. Shake	Loss IIB: 0.6	Solvent Test 4-hr. Shake	Loss S	Loss Solvent nk. Test 0.0	Loss	Loss Solvent	Loss mg.	Total Loss 1.1
	103.1	2 2 2	0.5	=	9.0	E	0.1	1	1	I	1	1.1
	101.7	2	0.7	=	9.0	2	0.0	!	1	1	1	1:3
	102.1	r	0.5	:	9.0	•	0.0	i	1	1	1	1.2
	104.3	=	9.0	I	9.0	=	0.0	1	1	:	1	1.2

## RPL-TDR-64-38 DISTRIBUTION LIST

U.S. Department of the Interior 1 Bureau of Mines 4800 Forbes Avenue Pittsburgh, Pennsylvania 15213 Attn: M.M. Dolinar, Repts Librarian	ivat'l Aeronautics & Space Admin l Washington, D. C. 20546 Attn: Office of Technical Information & Education Programs, Code ETL
Explosives Research Center	Scientific & Tech. Info. Facility 2 P.O. Box 5700
Central Intelligence Agency 1 2430 E Street, N.W. Washington, D.C. 20505	Bethesda, Maryland 20014 Attn: NASA Representative
Attn: OCD, Standard Dist	Nat'l Aeronautics & Space Admin. 1 Washington, D. C. 20546
Office of the Director of Defense 1 Research and Engineering	Attn: R. W. Ziem (RPS)
Washington, D. C. 20301 Attn: W. E. Sheehan, Office of Asst. Dir. (Chemical Technology)	Nat'l Aeronautics & Space Admin. l Goddard Space Flight Center Greenbelt, Maryland 20771 Attn: Library
Nat'l Aeronautics & Space Admin. 1 Lewis Research Center 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Library	Defense Documentation Center 20 Cameron Station Alexandria, Virginia 22314
Nat'l Aeronautics & Space Admin. 1 Launch Operations Center Cocoa Beach, Florida 32931 Attn: Library	Air Force Systems Command 1 Attn: SCTAP Andrews AFB Washington, D. C. 20331
Nat'l Aeronautics & Space Admin. Manned Spacecraft Center P.O. Box 1537 Houston, Texas 77001	Arnold Eng. Development Center 1 Attn: AEOIM Air Force Systems Command Tullahoma, Tennessee 37389
Attn: Library	AFRPL (RPR) 1 Edwards, California 93523
George C. Marshall Space Flight Cente intsville, Alabama 35800 Attn: Library	AFRPL (RPM) l Edwards, California 93523
Langley Research Center Langley Air Force Base	AFRPL (RPCC) 60 Edwards, California 93523

AFFTC (FTAT-2) Edwards AFB, California 93523	1	Commanding Officer Picatinny Arsenal	1
Office of Research Analysis Attn: RRRT Holloman AFB, New Mexico 88330	1	Liquid Rocket Propulsion Laborato Dover, New Jersey 07801 Attn: Technical Library	ry
Air Force Missile Test Center Attn: MTBAT Patrick Air Force Base, Florida 329	1 925	U.S. Army Missile Command Redstone Scientific Info. Center Redstone Arsenal, Alabama 35808 Attn: Chief, Document Section	4
Wright-Patterson AFB, Ohio 45433 Attn: AFML(MAAE)	1	Commanding General White Sands Missile Range New Mexico 88002	1
Commanding Officer Ballistic Research Laboratories	1	Attn: Technical Library	
Aberdeen Proving Ground, Maryland Attn: AMXBR-1 21005		Bureau of Naval Weapons Department of the Navy Washington, D. C. 20360 Attn: DLI-3	2
Commanding Officer Harry Diamond Laboratories	1	Bureau of Naval Weapons	,
Washington, D. C. 20438 Attn: AMXDO-TIB		Department of the Navy Washington D. C. 20360 Attn: RMMP-2	2
Commanding Officer U.S. Army Research Office (Durham)	!		
Box CM, Duke Station Durham, North Carolina 27706		Bureau of Naval Weapons Department of the Navy Washington, D.C. 20360 Attn: RMMP-3	1
Commanding Officer	1	Attn: KWIMP-3	
U.S. Army Chemical Research & Development Laboratories Edgewood Arsenal, Maryland 21010 Attn: Chief, Physicochemical Research Division		Bureau of Naval Weapons Department of the Navy Washington, D. C. 20360 Attn: RMMP-4	1
Commanding Officer Frankford Arsenal Philadelphia, Pennsylvania 19137	1	Bureau of Naval Weapons Department of the Navy Washington, D.C. 20360 Attn: RRRE-6	1
Attn: Propellant & Explosives S tion, 1331	. 1	Commander U. S. Naval Missile Center	2
Commanding Officer Picatinny Arsenal Dover, New Jersey 07801 Attn: Library		Point Mugu, California 93041 Attn: Technical Library	

Commander U.S. Naval Ordnance Laboratory White Oak Silver Spring, Maryland 20910 Attn: Library	2	Aerojet-General Corporation 11711 South Woodruff Avenue Downey, California 90241 Attn: Florence Walsh, Librarian	1
Commander U.S. Naval Ordnance Test Station China Lake, California 93557 Attn: Code 45	5	Aerojet-General Corporation P.O. Box 1947 Sacramento, California 95809 Attn: Technical Information Office	3 e
Commander (Code 753) U.S. Naval Ordnance Test Station China Lake, California 93557 Attn: Technical Library	2	Aeronutronic Div. Philco Corp. Ford Road Newport Beach, California 92600 Attn: Dr. L.H. Linder, Manager Technical Information Dept.	1
Superintendent U.S. Naval Postgraduate School Naval Academy Montercy, California	1	Acroprojects, Inc. 310 East Rosedale Avenue West Chester, Pennsylvania 19380 Attn: C.D. McKinney	1
Commanding Officer U.S. Naval Propellant Plant Indian Head, Maryland 20640 Attn: Technical Library	1	Aerospace Corporation P. O. Box 95085 Los Angeles, California 90045 Attn: Library-Documents	2
Commanding Officer Office of Naval Research 1030 E. Green Street Pasadena, California 91101	1	Allied Chemical Corporation General Chemical Division Research Lab., P.O. Box 405 Morristown, New Jersey 07960 Attn: L.J. Wiltrakis, Security Off	l ice:
Director (Code 6180) U.S. Navai Research Lab. Washington, D.C. 20390 Attn: H.W. Carhart	1	Amcel Propulsion Company Box 3049 Asheville, North Carolina 28802	1
Director Special Projects Office Department of the Navy Washington, D. C. 20360	1	American Cyanamid Company 1937 W. Main Street Stamford, Connecticut 06902 Attn: Security Officer	1
Commanding Officer .S. Naval Underwater Ordnance St Newport Rhode Island 02844 Attn: W.W. Bartlett		IIT Research Institute Technology Center	1
Acrojet-General Corporation P.O. Box 296 Azusa, California 91703 Attn: Librarian	1	Chicago, Illinois 60616 Attn: C.K. Hersh, Chemistry Division	۰

Distribution p. 4	
ARO, Inc.  Arnold Engrg. Dev. Center  Arnold AF Station, Tennessee 37389  Attn: Dr. B. H. Goethert  Director of Engrg.	The Dow Chemical Company 1 Security Section Box 31 Midland, Michigan 48641 Attn: Dr. R.S. Karpiuk, 1710 Bldg.
Atlantic Research Corporation 2 Shirley Highway & Edsall Road Alexandria, Virginia 22314 Attn: Library	E.I. duPont deNemours & Co. 1 Eastern Laboratory Gibbstown, New Jersey 08027 Attn: Mrs. Alice R. Steward
Atlantic Research Corporation I Western Division 209 Easy Street El Monte, California 91731 Attn: H. Niederman	Esso Research & Engineering Co. 1 Special Projects Unit P.O. Box 8 Linden, New Jersey 07036 Attn: Mr. D. L. Baeder
Battelle Memorial Institute 1 505 King Avenue Columbus, Ohio 43201 Attn: Report Library, Room 6A	General Dynamics/Astronautics 1 P. O. Box 1128 San Diego, California 92112 Attn: Library & Information Services (128-00)
Bell Aerosystems 1 Box 1 Buffalo, New York 14205 Attn: T. Reinhardt	General Electric Company 1 Cincinnati, Ohio 45215 Attn: Technical Information Center
The Boeing Company Aerospace Division P.O. Box 3707 Seattle, Washington 98124 Attn: R.R. Barber, Lib. Ut. Ch.	Hercules Powder Company 1 Allegany Ballistics Laboratory P.O. Box 210 Cumberland, Maryland 21501 Attn: Library
Chemical Propulsion Information 3 Agency Applied Physics Laboratory 8621 Georgia Avenue Silver Spring, Maryland 20910	Hercules Powder Company 1 Research Center Wilmington, Delaware 19800 Attn: Dr. Herman Skolnik, Manager Technical Information Division
Defense Metals Information Center   Batte <sup>11</sup>   Memorial Institute   505 King Avenue   Columbus, Ohio   42301	Jet Propulsion Laboratory 1 4800 Oak Grove Drive Pasadena, California 91103 Attn: Library (TDS) Mr. N. E. Devereux
Douglas Aircraft Co., Inc. 1 Santa Monica Division Santa Monica, California 90405 Attn: Mr. J. L. Waisman	Lockheed Propulsion Company 3 P.O. Box 111 Redlands, California 92374 Attn: Miss Belle Berlad, Librarian

Marquardt Corp. 16555 Saticoy Street Box 2013 - South Annex Van Nuys, Galifornia 91404	1	Texaco Experiment Incorporated P.O. Box 1-T Richmond, Virginia 23202 Attn: Librarian	1
Martin Company Baltimore, Maryland 21203 Attn: Science - Technology Library - Mail 398	1	United Technology Center P.O. Box 358 Sunnyvale, California 94088 Attn: Librarian	1
Minnesota Mining & Manufacturing C 900 Bush Avenue St. Paul, Minnesota 55106 Attn: J.D. Ross VIA: H.C. Zeman Security Administrator	o. 2	Thiokol Chemical Corporation Alpha Division, Huntsville Plant Huntsville, Alabama 35800 Attn: Technical Director	1
Monsanto Research Corporation Boston Labs., Everett Station Boston, Massachusetts 02149 Attn: Library	1	Thiokol Chemical Corporation Alpha Division Space Booster Plant Brunswick, Georgia 31500	1
New York University Dept. of Chemical Engineering New York, New York 10053 Attn: P.F. Winternitz	1	Thiokol Chemical Corporation Elkton Division Elkton, Maryland 21921 Attn: Librarian	1
North American Aviation, Inc. Space & Information Systems Div. 12214 Lakewood Boulevard	1	Thiokol Chemical Corporation Reaction Motors Division Denville, New Jersey 07834 Attn: Librarian	1
Nocketdyne 6633 Canoga Avenue Canoga Park, California 91304	3	Thiokol Chemical Corporation Rocket Operations Center P.O. Box 1640 Ogden, Utah 84401 Attn: Librarian	1
Attn: Library, Dept, 596-306  Rohm & Haas Company Redstone Arsenal Research Division Huntsville, Alabama 35808 Attn: Librarian	1	Thiokol Chemical Corporation Wasatch Division P.O. Box 524 Brigham City, Utah 84302 Attn: Library Section	2
Space Technology Laboratory, Inc. 1 Space Park Redondo Beach, California 90200 Attn: STL Tech. Lib. Doc. Acquisitions.	2	Union Carbide Chemicals Co. P.O. Box 8584 (Technical Center) South Charleston, W. Virginia 2530 Attn: Dr. H. W. Schulz	1

Chem. Research & Dev. Center United Aircraft Corporation FMC Corporation Pratt & Whitney Fla. Res. & Dev. Ctr. P.O. Box 8 P.O. Box 2691 Princeton, New Jersey 08540 W. Palm Beach, Florida 33402 Attn: Security Officer Attn: Library 1 United Aircrast Corporation Ethyl Corporation P.O. Box 3091 Corporation Library Istrouma Branch 400 Main Street Baton Rouge, Louisiana 70805 East Hartford, Connecticut 06118 Attn: Dr. David Rix 1 Hynes Research Corporation 308 Bon Air Avenue General Electric Company Durham, North Carolina 27704 Apollo Support Department P.O. Box 2500 Walter Kidde Company Daytona Beach, Florida 32015 675 Main Street British Defence Staff\* Belleville, New Jersey 07109 Attn: Security Librarian British Embassy 3100 Massachusetts Avenue Washington, D.C. 20008 Olin Mathieson Chemical Corp. Attn: Scientific Information Officer Research Library 1-K-3 275 Winchester Avenue New Haven, Connecticut 06511 Defence Research Member \* Attn: Mail Control Room Canadian Joint Staff (W) 2450 Massachusetts Avenue Miss Laura M. Kajuti Washington, D.C. 20008 Pennsalt Chemicals Corporation Technological Center Headquarters, U.S. Air Force 900 First Avenue Washington, D.C. King of Prussia, Pennsylvania 19406 20330 Attn: AFRSTD Purdue University Lalayette, Indiana 47907 AFRPL (RPCL) Attn: M.J. Zucrow Edwards, California 93523 Shell Development Company. Callery Chemical Company Research & Development 1400 53rd Street Emeryville, California 94608 Callery, Pennsylvania 16024 Attn: Document Control

\* Reports prepared under a Department of the Air Force contract or project will be transmitted VIA:

Headquarters
Air Force Systems Command
Attn; SCS-41(3-1833)
Andrews Air Force Base
Washington, D. C. 20331

Southwest Research Institute 1
Department of Structural Research
8500 Culebra Road
San Antonia, Texas 78223
Attn: Dr. Robert C. DeHart, Director

Texaco, Inc. 1
P. O. Box 509
Beacon, New York 12508
Attn: Dr. R. E. Conary, Manager